

A Direct ALE Approach for MultiMaterial Flows on Unstructured Grids

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A direct ALE method for multi-material flows on unstructured grids is under development

- Based on the generalization of a single-material Godunov scheme to multiple materials
- Key features
 - Direct solution of the multi-material ALE hydro equations
 - Approximate Riemann solvers in moving reference frame
 - Mesh velocity based on local fluid velocity with smoothing
 - Algebraic treatment of volume fraction equation
- Preliminary results on initial test problems have been obtained
 - Multiple interface methods under investigation
 - Further testing/validation is in progress

The ALE algorithm solves the flux-conservative Euler equations written for a moving reference frame

- The velocity of the reference frame is denoted by w_i with Eulerian and Lagrangian treated as limits

$$\frac{1}{V} \left[\frac{\partial(V\rho)}{\partial t} \right]_w + \frac{\partial}{\partial x_i} [\rho(u_i - w_i)] = 0$$

$$\frac{1}{V} \left[\frac{\partial(V\rho u_i)}{\partial t} \right]_w + \frac{\partial}{\partial x_j} [\rho u_i (u_j - w_j) + \delta_{ij} p] = 0$$

Lagrangian limit: $w_i = u_i$
Eulerian limit: $w_i = 0$

$$\frac{1}{V} \left[\frac{\partial(V\rho E)}{\partial t} \right]_w + \frac{\partial}{\partial x_i} [\rho E (u_i - w_i) + u_i p] = 0$$

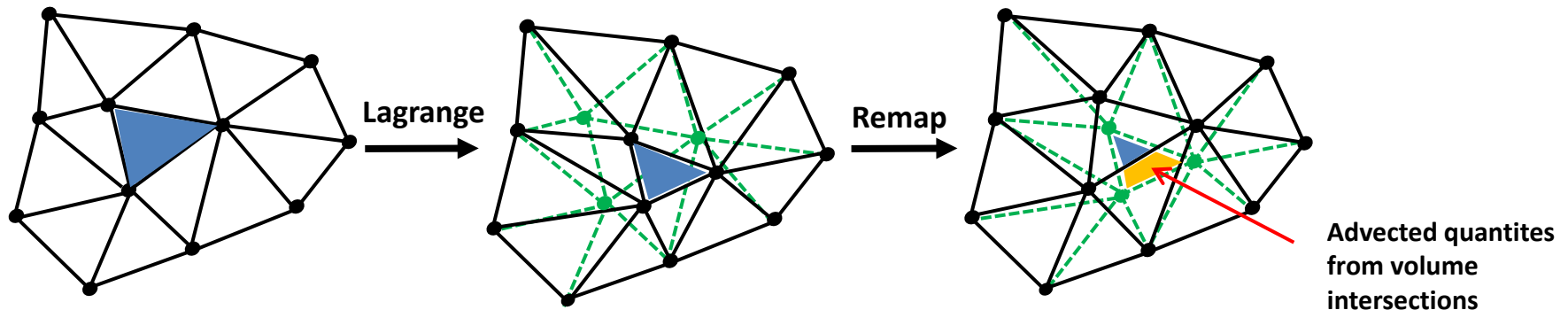
- Or in a more compact vector notation

$$\frac{1}{V} \frac{\partial(V\bar{U})}{\partial t} + \frac{\partial \bar{F}_j}{\partial x_j} = 0 \quad \bar{U} = \begin{pmatrix} \rho \\ \rho u_i \\ \rho E \end{pmatrix} \quad \bar{F}_j = \begin{pmatrix} \rho(u - w)_j \\ \rho u_i (u - w)_j + \delta_{ij} p \\ \rho E (u - w)_j + u_j p \end{pmatrix}$$

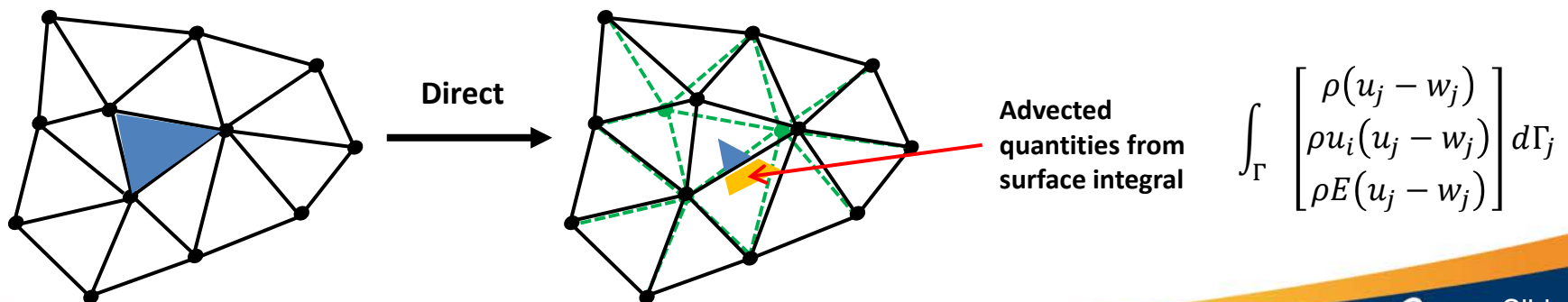
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The hydro equations are solved directly in the moving reference frame with no Lagrange step

- Lagrange-plus-remap: Mesh moves with fluid in Lagrange step, then relative to the fluid in remap step



- Direct ALE: Mesh moves independently of fluid at its own velocity



The multi-material extension adds equations for the volume fraction α_k

- Hydro equations are written for a material k
 - Separate thermodynamic state for each material in a control volume
 - Single fluid velocity and momentum equation applied as a closure

$$\frac{1}{V} \frac{\partial(V\bar{U})}{\partial t} + \frac{\partial \bar{F}_j}{\partial x_j} = \bar{S}$$

$$\bar{U} = \begin{pmatrix} \alpha_k \\ \rho_k \alpha_k \\ \rho_k \alpha_k u_i \\ \rho_k \alpha_k E_k \end{pmatrix} \quad \bar{F}_j = \begin{bmatrix} \alpha_k (u-w)_j \\ \rho_k \alpha_k (u-w)_j \\ \rho_k u_i (u-w)_j + \delta_{ij} p \\ \rho_k \alpha_k E_k (u-w)_j + \alpha_k u_j p_k \end{bmatrix} \quad \bar{S} = \begin{pmatrix} \alpha_k \frac{\partial u_i}{\partial x_i} \\ 0 \\ 0 \\ p_k u_i \frac{\partial \alpha_k}{\partial x_i} \end{pmatrix}$$

Advection is computed directly as part of the system of equations instead of from volume intersections

- Eulerian form of volume fraction equation:

$$\frac{\partial \alpha_k}{\partial t} + u_i \frac{\partial \alpha_k}{\partial x_i} = 0$$

- Transform to moving reference frame:

$$\frac{\partial}{\partial t_w} = \frac{\partial}{\partial t} + w_i \frac{\partial}{\partial x_i} \rightarrow \frac{\partial \alpha_k}{\partial t_w} + (u_i - w_i) \frac{\partial \alpha_k}{\partial x_i} = 0$$

- Rewrite in flux-conservative form (and dropping the w subscript):

$$\frac{1}{V} \frac{\partial (V \alpha_k)}{\partial t} + \frac{\partial}{\partial x_i} [\alpha_k (u_i - w_i)] = \alpha_k \frac{\partial u_i}{\partial x_i}$$

The ALE equations are discretized with an edge-based Finite Element method for linear tetrahedra

- Semi-discrete weak Galerkin formulation with element-based stencil :

$$\frac{d}{dt}(V\bar{U})^v + \sum_{\Omega_h \in v} \left\{ \sum_{w \in \Omega_h} \int_{\Omega_h} N^v \frac{\partial N^w}{\partial x_j} d\Omega_h \cdot (\bar{F}_j)^w \right\} = 0$$

sum over elements Ω_h
surrounding node v

sum over nodes w
contained in Ω_h

- Algebraically equivalent edge-based stencil:

$$\frac{d}{dt}(V\bar{U})^v + \sum_{vw \in v} \left\{ \left[\frac{1}{2} \sum_{\Omega_h \in vw} \int_{\Omega_h} \left(\frac{\partial N^v}{\partial x_j} N^w - N^v \frac{\partial N^w}{\partial x_j} \right) d\Omega_h \right] \cdot \underbrace{\left[(\bar{F}_j)^v + (\bar{F}_j)^w \right]}_{\text{Replaced with flux from Riemann solver}} \right\} = 0$$

sum over edges vw
surrounding node v

sum over elements Ω_h
surrounding vw

Replaced with flux from
Riemann solver

Consistency and conservation are ensured *by construction* for all choices of mesh velocity

- Semi-discrete equation for mesh point v is written with a single Riemann flux \bar{F}_i^{vw} on each edge as

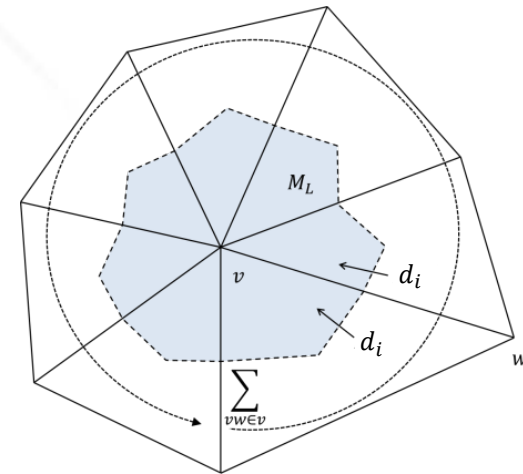
$$\frac{d}{dt}(V\bar{U})^v + \sum_{vw \in v} D_i^{vw} \bar{F}_i^{vw} = 0$$

$$D_i^{vw} = \frac{1}{2} \sum_{\Omega_h \in vw} \int_{\Omega_h} \left(\frac{\partial N^v}{\partial x_j} N^w - N^v \frac{\partial N^w}{\partial x_j} \right) d\Omega_h = \frac{1}{2} \sum_{\Omega_h \in vw} d_i^{vw, \Omega_h}$$

- Consistency and conservation statements:

$$\lim_{\Omega_h \rightarrow 0} \left[\frac{d}{dt}(V\bar{U})^v + \sum_{vw \in v} D_i^{vw} \bar{F}_i^{vw} \right] = \int_{\Omega} \frac{1}{V} \frac{\partial(V\bar{U})}{\partial t} d\Omega + \int_{\Gamma} \bar{F}_i d\Gamma_i \quad \forall w_i$$

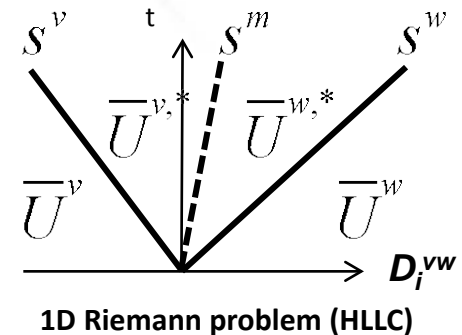
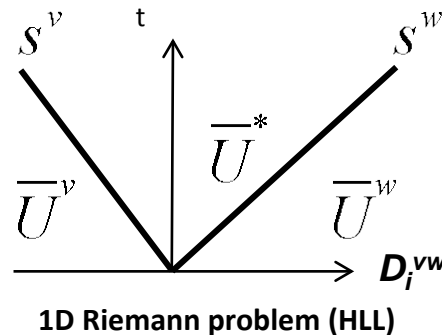
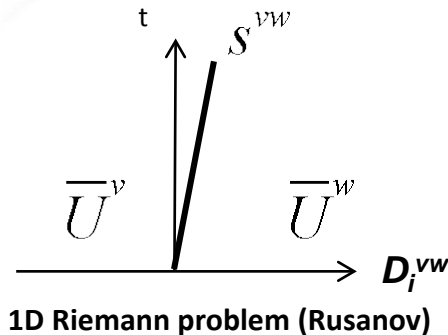
$$D_i^{wv} \bar{F}_i^{vw} = -D_i^{vw} \bar{F}_i^{vw} \quad \forall w_i$$



Edge-based stencil in 2D

A directionally unsplit flux on each edge is computed from an approximate Riemann problem along D_i^{vw}

- Rusanov, HLL, and HLLC schemes have been implemented



- Rusanov equations (shown for simplicity):

$$s^v = \left| \left(u_i^v - w_i^v \right) \frac{D_i^{vw}}{|D_i^{vw}|} \right| + c^v; \quad s^w = \left| \left(u_i^w - w_i^w \right) \frac{D_i^{vw}}{|D_i^{vw}|} \right| + c^w; \quad s^{vw} = \max(s^v, s^w)$$

$$D_i^{vw} \bar{F}_i^{vw} = D_i^{vw} \left(\bar{F}_i^v + \bar{F}_i^w \right) + |D_i^{vw}| s^{vw} \left(\bar{U}^w - \bar{U}^v \right)$$

A MUSCL reconstruction on primitive variables is used to obtain a second-order upwind method

- Given a field variable f with nodal values f^v and f^w , the reconstruction on edge vw is written as

$$\tilde{f}^v = f^v + \frac{1}{4} \left[(1+k) \phi(r^v) \delta_1 + (1-k) \phi\left(\frac{1}{r^v}\right) \delta_2 \right]$$

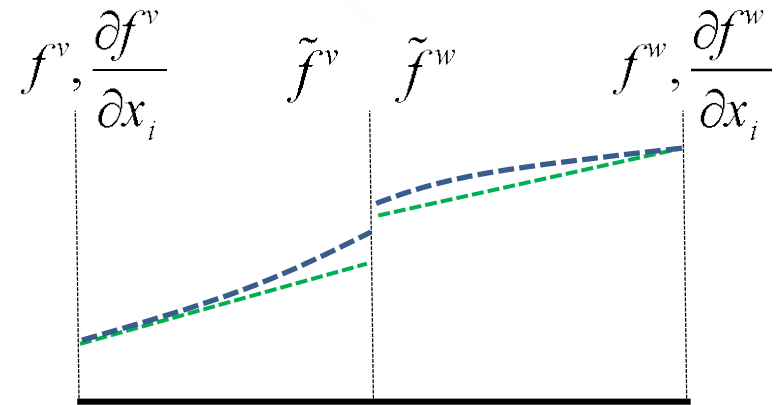
$$\tilde{f}^w = f^w - \frac{1}{4} \left[(1+k) \phi(r^w) \delta_3 + (1-k) \phi\left(\frac{1}{r^w}\right) \delta_2 \right]$$

$$\delta_1 = 2x_i^{vw} \frac{\partial f^v}{\partial x_i} - \delta_2$$

$$\delta_2 = f^w - f^v$$

$$\delta_3 = 2x_i^{vw} \frac{\partial f^w}{\partial x_i} - \delta_2$$

ϕ is any symmetric TVD limiter



Piecewise linear: $k = -1$

Piecewise parabolic: $k = 1/3$

The semi-discrete ALE equations are integrated in time with an explicit multi-stage scheme

- Difference formula:

$$(V\bar{U})^{v,0} = (V\bar{U})^{v,n}$$

$$\alpha^k = \frac{1}{1+m-k}$$

$$(V\bar{U})^{v,k} = (V\bar{U})^{v,0} - \alpha^k \Delta t \left(\sum_{vw \in v} D_i^{vw} \bar{F}_i^{vw} \right)^{k-1}$$

$$(V\bar{U})^{v,n+1} = (V\bar{U})^{v,m}$$

where m is the total number of stages and $1 \leq k \leq m$

- Same scheme used for mesh motion

- Summary of ALE step

1. Compute mesh velocity
2. Compute gradients
3. On each edge:
 - a. Compute MUSCL reconstruction
 - b. Compute flux from Riemann solver
4. Sum flux increments to points
5. Update $V\bar{U}$ to $n+1$
6. Update geometry (x_i, V, D_i^{vw}) to $n+1$
7. Update \bar{U} to $n+1$
8. Update p and c from EOS

The mesh motion is computed from a physics-based mesh velocity rather than a quality-based mesh relaxer

- Smoothed Lagrangian approach: Set $w_i = u_i$ then iterate on the linear system

$$\nabla^2(kw_i) = 0 \quad k = c_1 \max \left(0, 1 - c_2 \frac{\omega_i}{\|\omega_i\|_\infty} \right)$$

- Helmholtz approach: Set w_i equal to the irrotational component of u_i as computed from a Helmholtz decomposition

$$u_i = u_{i,rot} + u_{i,irrot} = u_{i,rot} + \frac{\partial \phi}{\partial x_i}$$

The scalar potential is computed from the solution to the Poisson equation

$$\frac{\partial^2 \phi}{\partial x_i^2} = \frac{\partial u_i}{\partial x_i} \rightarrow w_i = \frac{\partial \phi}{\partial x_i}$$

Key features:

1. Mesh velocity is independent of local mesh size
2. Mesh is stationary where the fluid is stationary
3. High-vorticity regions behave more Eulerian
4. Low-vorticity regions behave more Lagrangian

A mesh force model similar to Lagrangian Q models is applied to the mesh velocity for additional robustness

- The mesh velocity at each node v is evolved according to

$$w_i^{v,k+1} = u_i^k + \frac{\alpha^k \Delta t}{V^{v,k}} F_i^{v,k}$$

- The pseudo-force F_i is computed as a surface integral of pseudo-pressures (recall that d_i is a surface area normal of a control volume facet)

$$F_i^{v} = \frac{1}{2} \sum_{vw \in v} \sum_{\Omega_h \in vw} d_i^{vw} q^{\Omega}$$

- The pseudo-pressure q is computed from VNR- and TQS-like terms

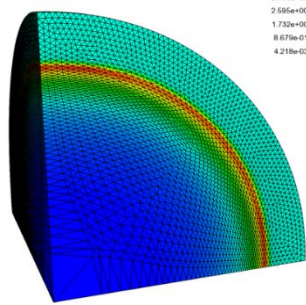
$$q = q^{\Delta u} + q^{\Delta V} = -\rho \left(b_1 c \Delta u + b_2 \Delta u^2 \right) - \rho b_3 c^2 \max \left(0, \frac{V^{t=0}}{V^k} - 1 \right)$$

VNR-like term

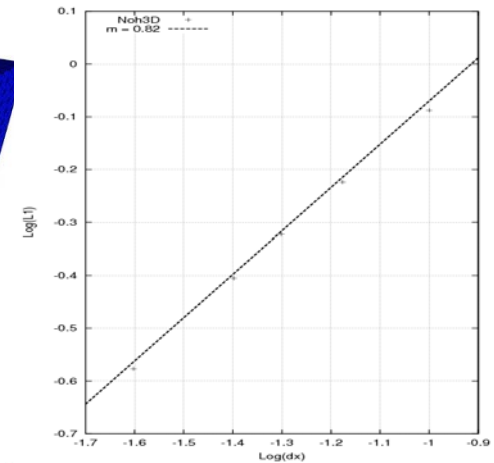
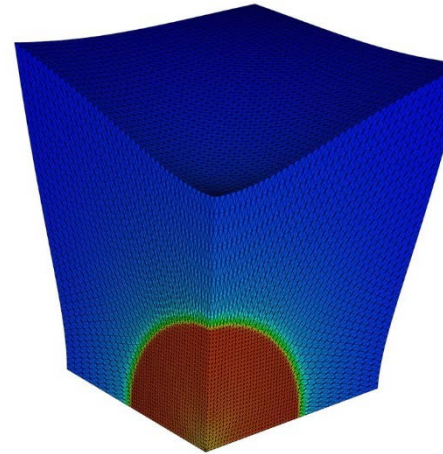
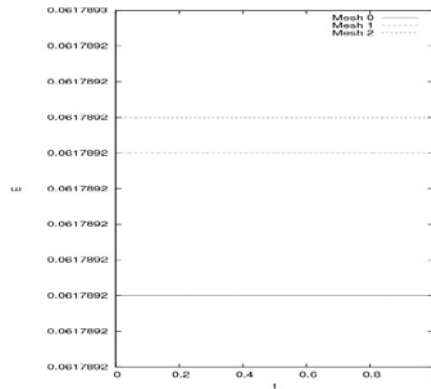
TQS-like term

The single-material ALE algorithm has been heavily verified on shock and shock-free problems

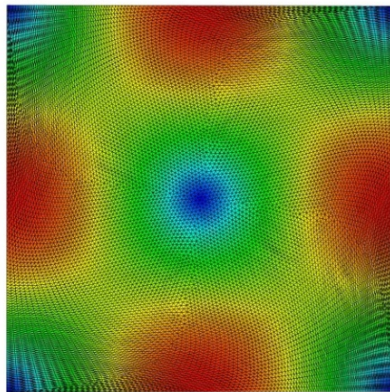
Time = 1.0000



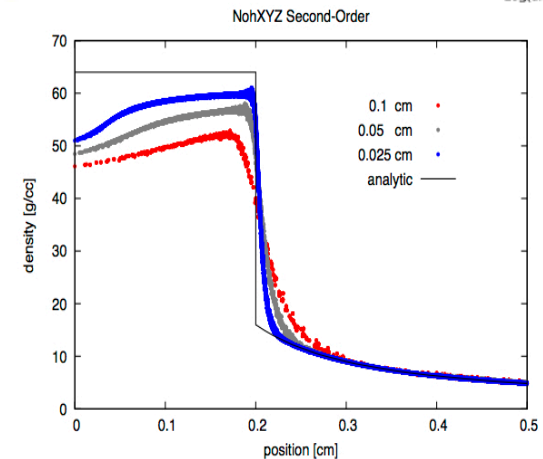
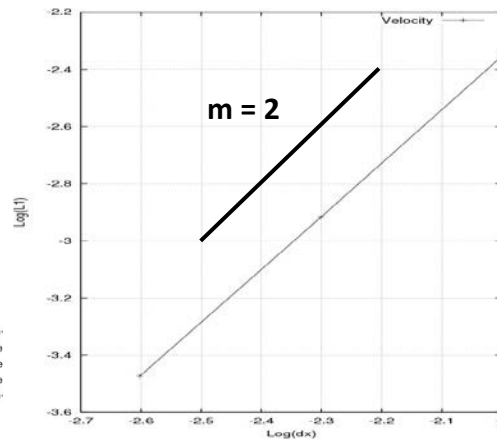
Sedov ALE



Time = 0.75



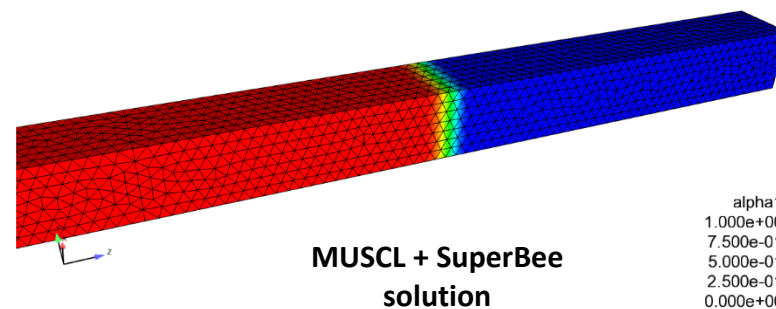
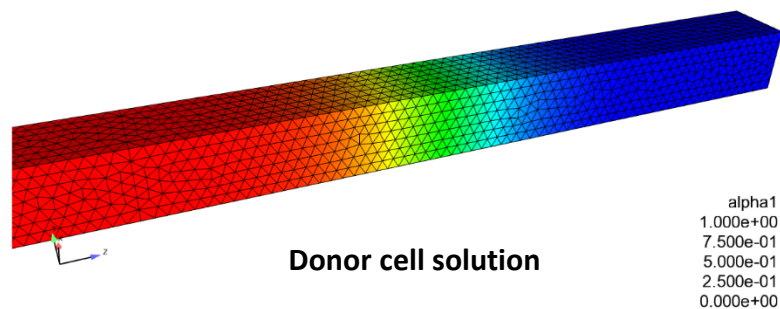
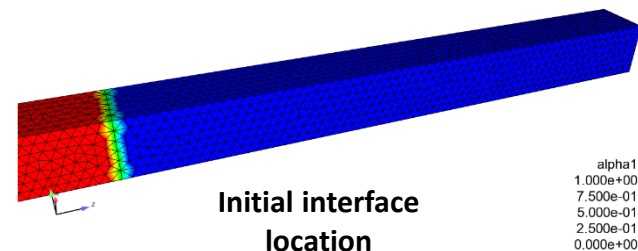
Taylor-Green ALE



Noh Lagrangian

The interface equation is initially being solved with a SuperBee limiter and Godunov flux

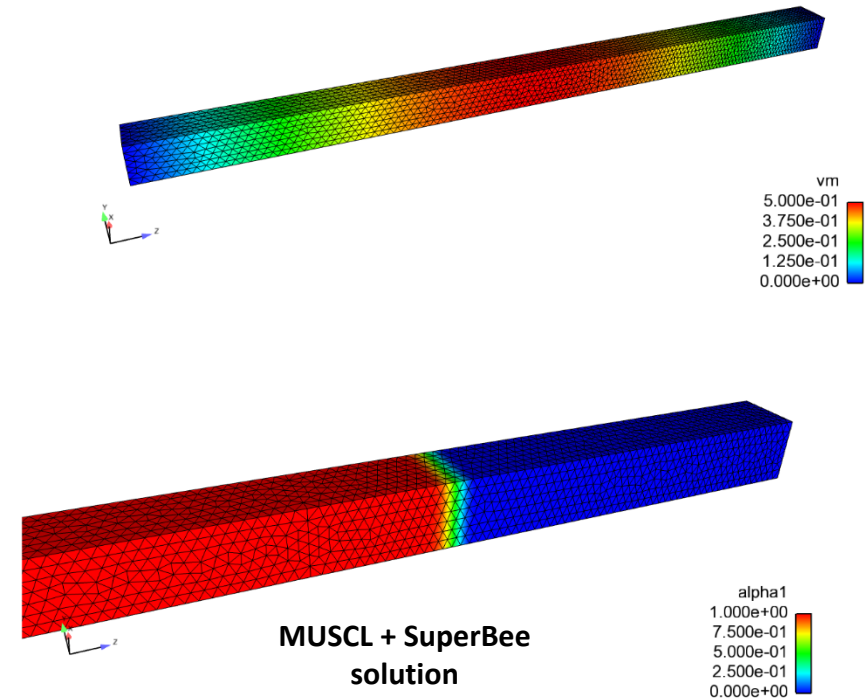
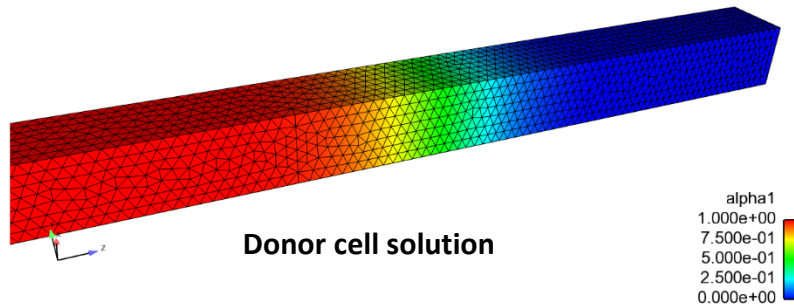
- Interface advection test problem (Eulerian)
 - Two materials, same state, constant u_i



- The Hyper C compressive limiter also is being tested
 - Caveat: the limit of zero numerical diffusion is perfect mesh imprinting! May not be the best choice on highly unstructured grids...

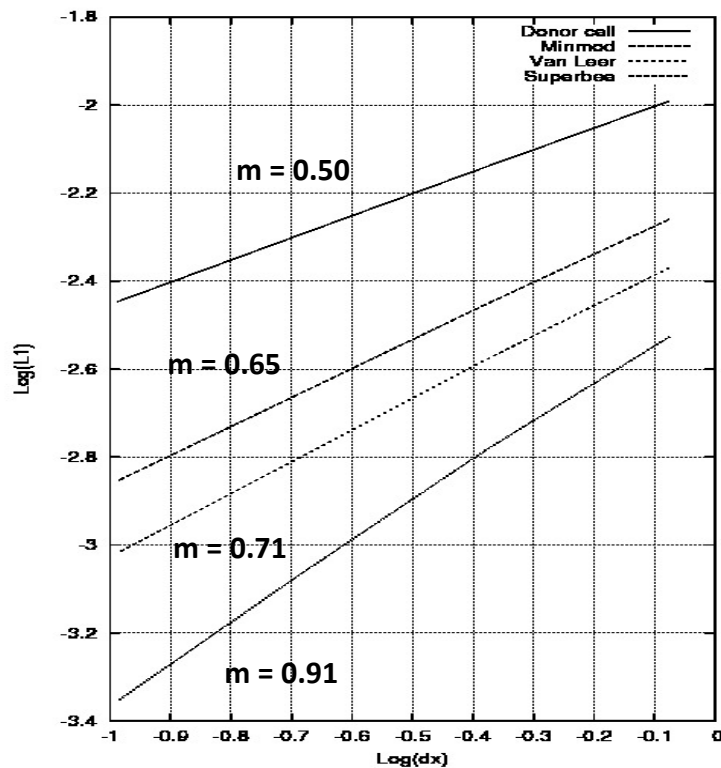
Similar results are obtained with ALE using a prescribed mesh velocity

- Interface advection test problem (ALE)
 - Sinusoidal mesh velocity along the length of the domain

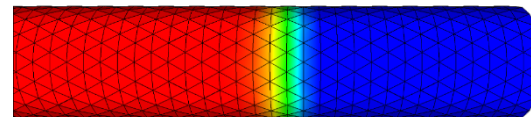
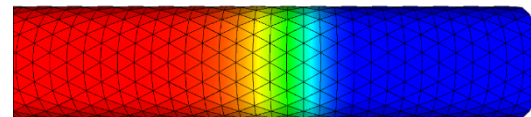
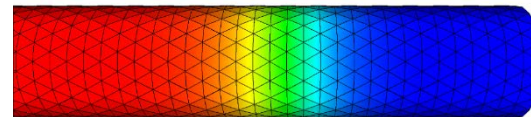
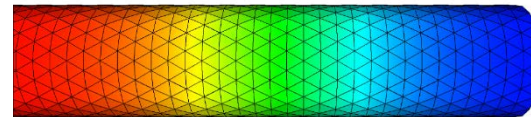


The SuperBee limiter obtains much closer to first-order convergence compared to other limiters

- Eulerian advection test + convergence study

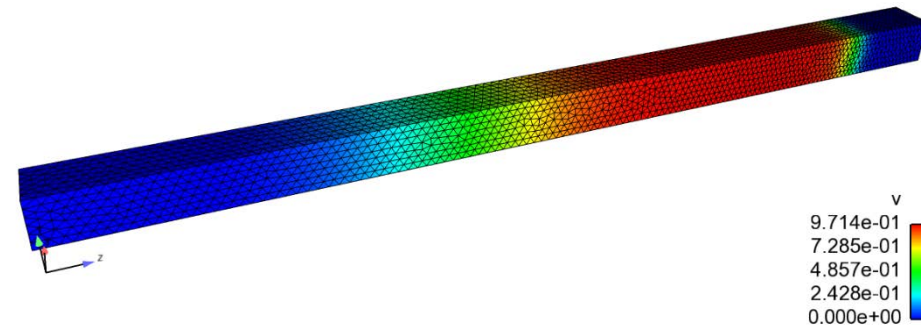
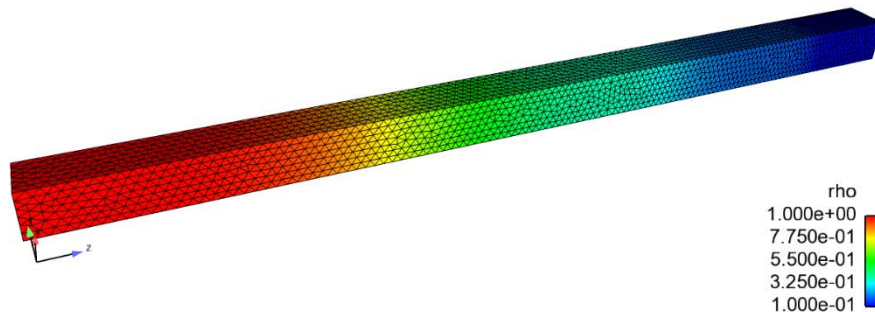
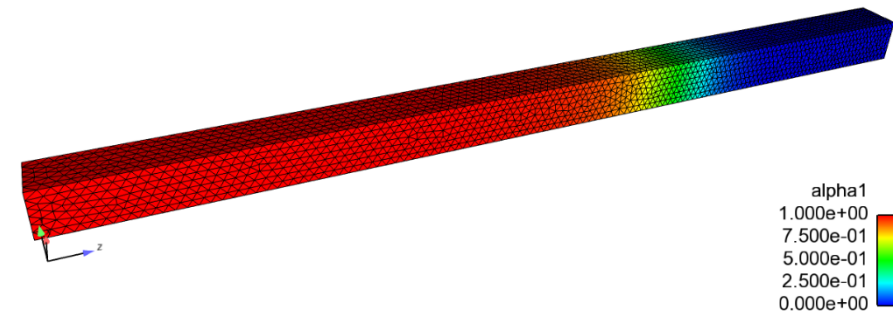
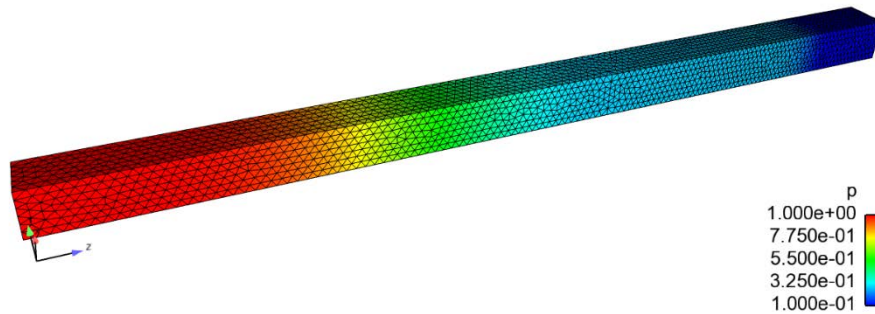


Time = 40.00



Initial donor cell results have been obtained on the shock tube problem

- Results are qualitatively correct (and diffuse as expected)
 - Smoothed Lagrangian mesh velocity



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